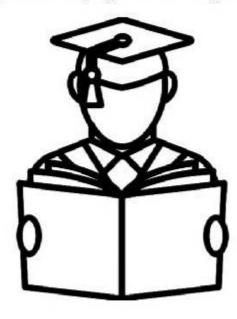


"I don't love studying. I hate studying. I like learning. Learning is beautiful."



"An investment in knowledge pays the best interest."

Hi, My Name is

<u>Chemical Engineering</u> for <u>GATE/IES</u> (MADE EASY)

6'	Matrices
©',	
•	Properties of determinants:
	(i) If two rows or columns of matrix are identical, then the
e,	determinant is zeo
6,	
e,	§ J 2
¢,	
©, ©,	
))	
© >	cit) If two rows or columns of matrix are interchanged then the
0)	sign of determinant changes.
0)	$\Delta = \begin{bmatrix} 1 & 2 & 3 \\ 3 & 1 & 1 \end{bmatrix} - \Delta = \begin{bmatrix} 1 & 0 & 2 \\ 1 & 0 & 2 \\ 2 & 1 & 3 \\ 1 & 3 & 1 \end{bmatrix}$
()	CAMBHOLOG VP 1 2 1 0 2
6)	$\Delta = 1 2 3 \qquad -\Delta = 2 1 3$
(C)	3 1 1 3 1
0)	
6)	(iii) If three rows or columns of matrix are interchanged then the
©) •)	sign of determinant is unaltered.
0	
- -	$ \Delta = \begin{pmatrix} 1 & 2 & 3 \end{pmatrix} = \begin{bmatrix} 2 & 1 & 1 \\ 2 & 1 & 1 \end{bmatrix} $
	$ \begin{array}{c} \Delta = \left(\begin{array}{ccc} 1 & 2 & 3 \\ 3 & 1 & 1 \end{array} \right) = \left(\begin{array}{c} 3 & 1 & 1 \\ \end{array} \right) \\ \end{array} $
e)	
¢ þ	av) In the determinant of matrix if any column containing sum
6 ₂	or orderence of two elements then it can be split into sum or
6 9	difference of two determinants.
C >	$ q q^2 q^3 + q q^2 q^3 + q q q^3 + q q q q q q q q q $
() »	$\begin{bmatrix} q & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} = \begin{bmatrix} q & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} = \begin{bmatrix} q & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$
C 3	$\begin{vmatrix} a & a^2 & a^3 + 1 \\ b & b^2 & b^3 + 1 \\ c & c^2 & c^3 + 1 \end{vmatrix} = \begin{vmatrix} q & a^2 & a^3 \\ b & b^2 & b^3 \\ c & c^2 & c^3 + 1 \end{vmatrix} = \begin{vmatrix} q & a^2 & a^3 \\ b & b^2 & b^3 \\ c & c^2 & c^3 \end{vmatrix} + \begin{vmatrix} a & a^2 & 1 \\ b & b^2 & 1 \\ c & c^2 & c^3 \end{vmatrix}$
09	$\begin{vmatrix} c & c^2 & c^3+1 \end{vmatrix} \begin{vmatrix} c & c^2 & c^3 \end{vmatrix} \begin{vmatrix} c & c^2 & J \end{vmatrix}$
()) ())	
¢ ·	
6	
• •	

	$A = \begin{bmatrix} \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\$	a b c d] d-bc	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	0		
	A =	$\begin{array}{ccc} 0_{11} & 0_{1} \\ 0_{21} & 0_{2} \\ \end{array}$	$ \begin{array}{ccc} p_{1} & \Omega_{13} \\ p_{22} & \Omega_{23} \\ p_{12} & \theta_{33} \end{array} $			م م م ا
E. C. C.	$\Delta = 0$			$\begin{bmatrix} 0 & 0 \\ 2 & 0 \\ 3 & 0 \\ 3 & 0 \\ 3 & 1 \end{bmatrix} + 0$	• :	(@ (© (©)
۰ ۴ د	tions tor solvin order of matrix		are always	s less than	one	() () ()
upper univ	lower triangula					٦
its e		er triongu	lor matrix are zero	and if all	elements triangular 1	
its d below	said to be lowe o the principal ix.	er triangu diogonal angular r	lor matrix are zero motrix =	and if all it is upper	elements triangular] = 18	

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$$\begin{bmatrix} 1 & 0 & -b & -b & -b \\ 1 & -b & -b^2 \\ 1 & -b & -b^2 \\ 1 & -c & -c^2 \end{bmatrix}$$

$$\begin{bmatrix} 0 & 0 & -b & -b^2 & -b^2 \\ 0 & -b & -c & -b^2 & -c^2 \\ 0 & -b & -c & -b^2 & -c^2 \\ 0 & -b & -c & -c^2 \end{bmatrix} = (0 - b) C (b - c) \begin{bmatrix} 0 & 1 & -c & +b \\ 0 & -j & -b & -c \\ 1 & c & -c^2 \end{bmatrix}$$

$$\begin{bmatrix} 0 & 0 & -b & -b^2 & -b^2 \\ 0 & -b & -c & -b^2 & -c^2 \\ 0 & -b & -c & -c^2 \end{bmatrix}$$

$$\begin{bmatrix} 0 & -b & -c & -c^2 \\ 0 & -b & -c & -c^2 \end{bmatrix}$$

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$$\begin{bmatrix} 0 & -b & -c & -c^2 \\ 0 & -b & -c & -c^2 \end{bmatrix}$$

$$\begin{bmatrix} 1 & 1 & -1 & 1 \\ 1 & -c & -c^2 \\ 0 & -c & -b & -c^2 \end{bmatrix}$$

$$\begin{bmatrix} 1 & 1 & -1 & 1 \\ 0 & -c & -c^2 \\ 0 & -c & -c^2 \end{bmatrix}$$

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$$\begin{bmatrix} 1 & 1 & -1 & 1 \\ 0 & -c & -c^2 \\ 0 & -c & -c^2 \end{bmatrix}$$

$$\begin{bmatrix} 1 & 1 & -1 & 1 \\ 0 & -c & -c^2 \\ 0 & -c & -c^2 \end{bmatrix}$$

$$\begin{bmatrix} 1 & 1 & -1 & -c & -c^2 \\ 0 & -c & -c^2 \\ 0 & -c & -c^2 \\ 0 & -c & -c^2 \end{bmatrix}$$

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$$\begin{array}{c} \mathbf{Q}_{c} \quad hno \ \text{ime} \ \text{actominant} \ \text{of} \\ \begin{array}{c} \left| \begin{array}{c} 1+\mathbf{Q}_{c} & 1 & 1 \\ 1 & 1+\mathbf{b}_{c} & 1 \\ 1 & 1+\mathbf{c}_{c} \\ \end{array} \right| \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \frac{\mathbf{R}_{1}}{\mathbf{B}_{c}} & \mathbf{R}_{2} \longrightarrow \frac{\mathbf{R}_{2}}{\mathbf{b}_{c}} & \mathbf{R}_{3} \longrightarrow \frac{\mathbf{R}_{3}}{\mathbf{b}_{c}} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \frac{\mathbf{R}_{1}}{\mathbf{b}_{c}} & \mathbf{R}_{1} \longrightarrow \frac{\mathbf{R}_{1}}{\mathbf{b}_{c}} & \mathbf{R}_{1} \\ \hline \mathbf{R}_{1} \longrightarrow \mathbf{R}_{1} + \left(\mathbf{R}_{2} + \mathbf{R}_{3}\right) \\ \end{array} \right| \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{1} + \left(\mathbf{R}_{2} + \mathbf{R}_{3}\right) \\ \end{array} \\ \end{array} \\ = \ abc \left(\left(1 + \frac{1}{\alpha} + \frac{1}{\mathbf{b}_{c}} + \frac{1}{\mathbf{c}_{c}}\right) \\ \hline \frac{1}{\mathbf{b}_{c}} & \frac{1+\frac{1}{\alpha}}{\mathbf{b}_{c}} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{1} + \left(\mathbf{R}_{2} + \mathbf{R}_{3}\right) \\ \end{array} \\ \end{array} \\ = \ abc \left(\left(1 + \frac{1}{\alpha} + \frac{1}{\mathbf{b}_{c}} + \frac{1}{\mathbf{c}_{c}}\right) \\ \hline \frac{1}{\mathbf{b}_{c}} & \frac{1+\frac{1}{\alpha}}{\mathbf{b}_{c}} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{1} + \left(\mathbf{R}_{2} + \mathbf{R}_{3}\right) \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{2} \longrightarrow \mathbf{R}_{1} + \left(\mathbf{R}_{2} + \mathbf{R}_{3}\right) \\ \end{array} \\ = \ abc \left(\left(1 + \frac{1}{\alpha} + \frac{1}{\mathbf{b}_{c}} + \frac{1}{\mathbf{c}_{c}}\right) \\ \end{array} \\ \begin{array}{c} \frac{1}{\mathbf{b}_{c}} & \frac{1+\frac{1}{\alpha}}{\mathbf{b}_{c}} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{1} + \left(\mathbf{R}_{2} + \mathbf{R}_{3}\right) \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{2} \longrightarrow \mathbf{R}_{2} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{1} \\ \end{array} \\ \begin{array}{c} \frac{1}{\mathbf{b}_{c}} & \frac{1}{\mathbf{b}_{c}} \\ \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{2} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{2} \longrightarrow \mathbf{R}_{2} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{2} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{2} \\ \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{2} \\ \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{2} \longrightarrow \mathbf{R}_{2} \\ \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{2} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{2} \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{2} \end{array} \\ \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{2} \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{2} \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{2} \longrightarrow \mathbf{R}_{2} \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \end{array} \\ \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \longrightarrow \mathbf{R}_{1} \end{array} \\ \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{2} \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{2} \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \end{array} \\ \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \end{array} \\ \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \end{array} \\ \end{array} \\ \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \end{array} \\ \end{array} \\ \end{array} \\ \begin{array}{c} \mathbf{R}_{1} \end{array} \\ \end{array} \\ \end{array} \\$$

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 $A = \begin{bmatrix} 1 & 2 \\ 3 & 4 \end{bmatrix}$ $A = \begin{bmatrix} 1 & 2 \\ 3 & 4 \end{bmatrix}$ $A = \begin{bmatrix} \cos \theta & -\sin \theta \\ \sin \theta & \cos \theta \end{bmatrix}$ $\overline{A} = \begin{bmatrix} \cos \theta & -\sin \theta \\ \sin \theta & \cos \theta \end{bmatrix}$ $\overline{A} = \begin{bmatrix} 1 & 2 \\ -3 & 1 \end{bmatrix}$ $A = \begin{bmatrix} \cos \theta & -\sin \theta \\ \sin \theta & \cos \theta \end{bmatrix}$

 $A = \begin{bmatrix} a & b \\ c & d \end{bmatrix}$

 $A^{-1} = \frac{\sigma d j \cdot A}{\Delta}$

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Inverse matrix exists only for non-singular matrix.

Adjoint of Higher order matrix is the transpose of the co-factor matrix.

Minor of element:

The minor of an element in square matrix is the determinant of square sub matrix in which the row and the column of particular element lies to be deleted.

$$A = \begin{vmatrix} 0 & 1 & 2 \\ 1 & 2 & 5 \\ 3 & 1 & 1 \end{vmatrix}$$

Minor of $a_{12} = \begin{vmatrix} 2 & 3 \\ 3 & 1 \end{vmatrix} = (1-9) = -8$

(o-factor of an element = $(-1)^{i+j}$ Minor of element

$$= (-1)^{-1} \times (-8)$$

i-no of row & j-no of column.

Ahemical reaction Engineering @ 2016 > to design a reaction vessel (reactor) is supe. of reactor (mode of operation) Volume / seje & reactor chemical reaction formation & breaking of new & old bonds cest ٢ 4 Homogeneous Heterogeneous Ì more than one phase Single phase reaction ٢ S-G siknt all Gras phase ં L-G skun For all higher phase *_____* Catalytic reaction ા પરિશ્વન હી અખલતા વડી વ્હુની હ steres phorostat fast - explasion JA SARAI, NEW DELHI-16 & ston - radioactine decay $1N_2 + 3H_2 \rightleftharpoons 2NH_3$ reverseble } exothermic) (Habee's procens) ٨ Contact reaction enoltermin ÷... 1, 3, 2 - Atoichiometrie co-efficients : co-efficients of reactants or products * atoichiometein cofficients of a chemical rek refresents moles, molecules, or volume (for gases such) the stoichiometric cofficient tells us about how the chemical deaction will proceed. (puts no restrictions on how much it should be taken) ٢ ٢ 3 ٢ ٢

H Conservation of mans @ is valid in Chemical p Akn 312 产 2NK3 IN2 + 20 miles 10 moles D £ = 0 enotes in OL CA. 2 moles 17 moles 9 moles t=ti (Innole reacted) time increases as net it ik haffers 34gm 6gm 28 gm mans 4 moles. 8 mols 14 mols $t=t_2$ 0 In reality no reaction goes to 0 completion, it () Stops before which ٢ is decided by therwodynamis. 12moles. mide ymoly t= tx 88 12+4/3 Completion: 0 mils t= ty 4-2/3 ୍ବିଚ The reactant which get conserved first is called the dimiting reactant, while the other one is called excess ٢ 0 ٨ reactant. **_** to find the limiting searchant, we have to assume that the 0 3 reaction goes to completion. ite find the limiting ouactant, we will denide the entit <u></u> (j?) initial number of males of seactants by their respective ۲ ۲ stoichiometric co-efficient,

The reactant which gives lesser value is limiting star reactant All this stoichiometric calculation in a reaction is done . ٢ ٧ on the basis of the limiting reactant ۲ Stoichiometric proportion - reactants are said to be in <u></u> ુ ۲ of the reactants is same as the ratio of the corresponding Ì stoichiometric cofficients. ġ 3 3 H2 - S ENKY N2 + ુ 10 mole 10 mole E=0 10 mole 30 mile t-- 0 E=0 · 1.0 Kg (no becomsi always it is done on 30 Kg 19 Trusis of mass male) we can convert it ento mol. 28 Kg/Kunol 2 kg/ Kunol If these are in stoichiometer proportion than both we get and ٢ at same time of either both can be limiting or none. reverbulie at it \rightleftharpoons cc +dD bB $K_{c} = \begin{bmatrix} c \end{bmatrix}_{e}^{c} \begin{bmatrix} o \end{bmatrix}_{e}^{d}$ (ale taken at equilibrium) $\left[A\right]_{e}^{a}\left[B\right]_{e}^{b}$

Grate 2017 8) The reversible reaction of tertiary bietyl alchahol of ethand to give ethyl terester butyl ether is given by. $TBA + ETOH \longrightarrow ETBE + H_{20}$ the equilibrium constant for this reaction is equal to and ۲ Initially 74 gm of TBA is mixed with 100 gm of ag soly Containing MGY. Chand by ht. Grien - MWTBA = 74 633 MW Etou = 46 03 ٨ MWETGE = 102 the mars of ETBE at eghin. + HLD EtBE TBA + ELOU FICE C The grin MEr. 100 mgrin t-0 \odot ෙ hour 216-046 0.54 1070 0 $\frac{54}{18} = 3 uol.$ ٢ 1 mole 1 viol E=0 1-2 3+2 t=teg. 1-2 X <u>.</u>... $\frac{2}{(1-2)(1-2)}$

 $i = 3x + x^2$ < In this question Ś noheme is constant $1 - 2x + x^{2}$ ٢ to in place of 1-2x +x2= 3x+x2 (i)) Conc^u we can take ٢ x - 0.2 mol, moles. 4 at equilibrium ٢ 0.2 mols of E.TBE Ð Q-2×102 Hen of ETRE = (i) 20.4 gm ٨ man of. H20 = 3.2×18 Ì 5 57.6 gm 3 Ś -> 'cC'+'db aA + bB+ convesion - it is only Ì defined only for searchants 8 and never for products Ŵ Convesión of a reactant A is denoted by XA ٧ X_A = moles of A reacted £ . utilale MAO - NA -> final lileft, ٧ ٢ ٢ $\frac{N_{A_0}}{1 - N_A} \rightarrow \text{for batch}$ M.G ۲ XA = 1 - "NA/t XA 3 NA0/t ٨ puola It can also be expressed as a ". KA = 1 - FA-for FAO ٢ Continue we should always use the ٢ fractional value of convesion 9 89

For reforting the final answer, we should read the question and according to that conversion chould be reported. $aA + bB \rightarrow cC + dD$ ED NAO NBO NCO NDO let us suppose conversion of A is known & it is XA (Ahere is limiting @ $N_A = N_{AO} (1 - X_A)$ under of A scencted = NAO XA B suacted = . (A suacted) $= \frac{b}{a} (NA_0 X_A)$ $N_B = N_{BO} - \frac{b}{a} (N_{AO} X_A)$ NC = NG+ C (NAOXA) $N_D = N_{D_D} + \frac{d}{a} (N_{A_D} X_A)$ relationship b/w KA + XB NB = NB (1 - XB) NBO-b-NBO(1-XB) $NBO - \frac{b}{a}(NAOXA) = NBO(1-XB)$ XB = b NAO XA

Fluid Mechanics 1719 Hel to the cross-sectional area of the object. Jormal forces It will change only the demensions not JFn Y the shapes -TYTETTET -> I'll to the cross rection area of the object. Shear forces It will sharinge only the shape not the demension of the object. F_S X' ા પરિક્રન હી સખલતા લ્કી લુહી है । eterst photostat JIA SARAI, NEW DELHI-16 Meb. No. 9818909565 It's a substance which deforms continuously under the action of shear forces i.e. fluid can't resists fluid shear. The continuous state of deformation is termed as flow. Matur, air- clastic behaniour is present in normal force sense and not in shear force rense. 20 has different clasticity. I diff deformation rate. - The velocity of all the particles is some I its Motion just vehange in displacement Velocity of all particles are different I its layer flow ٢ by layer displacement. ં ં Absolute Vacuum cannot be a system, but vacuum can be considered as a system.

System - 2ts an amount of matter whose physical characteritics are under observation / investigation. **1**..... Ŷ 0 Property - Its the physical characterities of a system which defines the state of the system eg - P, T, etc ۲ ٨ ۲ Intensive peroperty - They are those which are independent of mais eg- every specific property, @ . If a peroperty is defined at a point then its intensine peroperty I it doesn't need bulk to define it. ۲ Ì Entenne peroperty - The property will be entensine peroperty ۲ if it depend upon mars lor) if it requires $\langle \rangle$ ۲ a bulk to define it. eg- weight, volome, ٢ entropy, enthalpy. Physical peroperties of fluid () Mars density (9): density of reater is 1000 Kg/m³ only (9,4°C + 1 atm P ٢ ۲ ٨ density of Hg is 13600 kg/m3 ٢ only Q 25°C & latur P. ۲ Its the amount of mars of fluid occupied in unit notime ۲ ٨ at the given state of the fluid of = m 3 la lm3 **@**

Note density of water will always decrease no matter if you increase or decrease the temp (1e) why the sice floats on water. 2) Specific Volume (v): Specifie volume is defended for gaseous & not liquid. Its defined as volume occupied by unit mass of fluid. 68 Ŵ V= V The concept of specific volume is useful for compressible fluids (i.e.) gaseous because, gases dan occupy differentdifferent volume for the same amount of mais at a given state Specific weight (pleight density (w?): 12ts an amount of weight of the fluid occupied in whit 3 volume at the given state I in the given field (granitational field) $\dot{\omega} = \frac{W}{V} = N[m^3] = \frac{mg}{V} = \frac{g}{V}$ Specifie weight of water => 9.81×1000 => 9810 N/m³ ٢ Q 4.4°C} 3 · later P. 0 ÷.)

Q (Specific granity Its the ratio of mars density (or) weight density of **@** Unknown fluid to mars I weight I of some standard 63 (fluid or <u>w</u>. s.g = J Jst heavier denser For liquids, standard fluid is heater @ 4.4°C + latu P., ۲ For gases, stal fluids is air & fair = 1:23 kg @ 25°C, lat . for std. fluids, the specific gravity is ! Now, if for some fluid, the specific granity is less than 1. than that fluid will be lighter I less dense. compared to std fluid I it will float oner std fluid. eg- water & ice If the specific granity of unknown fluid is greater than 1. then that fluid is denser & hearnier than state fluid. t will rink donen in stal fluid. eg- Mercury 899

2 6 2 L of petrol of weight 14 N Cal Sp weight mans Is 3 sp. volume + sp. granity. 714.2857 g/m3 -> S.g. 3 <u>Gr</u>. 9, -> 14/9.8 3w 0:002 0.7142 3 g 3 714.2857 1 3 The petrol will float over Sp. reaght = $\frac{W}{V} = \frac{14}{0.002}$, 1000 N/m³ (5) <u>Bulk modulus of elasticity (k) + coefficient of</u> <u>compressibility (Bc)</u> -de top top Hooke's law - stress & strain +dw -dw Stress - 2ud order tensor quantity (duff. walne of diff. dir (diff. value of diff. directions) Mars zero order tensor is scalar Stress -> Induced quantity (same value in all Pressure -> Applied quantity directions) st order tensor -> Vector 8 V can induce stress by applying presence but not wice inersa. ંશુ 3 (mehren values are ٩ Moment of Inertia -> 2nd Older Tensor: diff. at diff. 8 mutually in directions

VN X EV dp x Er $k = \frac{d\rho}{\left(\frac{-dv}{v}\right)}$ $dp \propto - \frac{dv}{v}$ 6 $dp = -k \frac{dv}{v}$ 6 Under normal forces water acts as an elastic material. m = C m = g.V gdv + vdp = 0 $\frac{1}{k} = \frac{dP}{dP}\left(\frac{dS}{3}\right)$ dm = gdw = . - v ds 'k' Unit -> Pa (or) N/m2 $-\frac{dv}{V} = \frac{dy}{g}$ · 'k' value is greatenthan the pressure to be applied for a specific change in volume should be greater. . 'k' value is greater than the fluid behaves more clartic 'k' value is less than it for a greater ant of pressure also it will cause only less volume change -> From the enferiment & by using blooks's law - we can say that the preservie change will be perop. the change 0 in volume but opposite in nature (1.e) if <u>peressure 1</u> en

· 2016. Heat transfer Olsrof CP Arora T () . RC Sachdena Heat 3 11 DS Kumar 70°C 730°C | (4) 11. Domkundwad. , Rehen two bodies which see in contact, Water flows from than the transfer which occurs is called higher potential to * heat (the energy which flows by nietue lower potential till of temperature difference) they are at equal potential. * Janperture - thermal potential of system responsible for heat transfer 1 Unit of heat -88 1 gm of water X . 1°C - Cas 1 calorie i)) by of water × 1°C = 1 kilo Calorie 3 ।। परित्रम ही सफलता की क्हेजी है ।। 1 KCal टगेहारी PHOTOSTAT Ŵ JIA SARAI, NEW DELHI-16 Mob. No. 9818909565 I lb of water x 1°C -= 1 CHU 0 . Centigeade Heat Unit 3 1 Ib of water X 1°F B. ThU (Brotish Theemal Unit) **8**3 8 B.T.V or Btu ٨ Work & Heat (similar type of energy in transient state ٨ KNM K Kal 4.1868 KJ -== 1 KGal KJ & Kal o (. ,

K Cal is a fundamental chiet of Heat KJ is a derived that of Heat : * Modes of Heat teansfer 1) Conduction - 10 hes 2) Convection - 20 his 3) Radiation - 10 hrs. 6 hes exchangees I hes change in phase Conduction Plane Wall) assemptions 172 1 One-dimensional heat transfer @ Steady state heat transfer. (temperature does not change with itime) teansfer direction 3 & remains same (ut does not varry with itemperature) +T T, >T2 (7) No heat generation (5) Asothermal surfaces (No heat is retained in the System Fourier's law of heat Conduction Heat flor & Temperature gradient $\frac{g}{\Lambda} \propto \frac{d}{\Lambda}$ normal area to the direction of heat transfer

* <u>B</u> = -kdT A loh ٢ ----- Constant of proportionality 8 -> Thermal conductinity of material $g = -kA \frac{dT}{dx}$ $dn = -\frac{kA}{Q} dT$ Integrating, fdn = -kA fdT 8 89 $\chi_2 - \chi_1 = - \frac{kA}{N} (T_2 - T_1)$ Ś 3 $= \frac{kA}{Q}(T_1 - T_2)$ ۲ 1 $g = \frac{kA(T_1 - T_2)}{(x_2 - x_1)} = \frac{kA(T_1 - T_2)}{\Delta x}$ 8 ٧ $g = \frac{1}{\sqrt{1-T_2}} \frac{1}{\sqrt{\lambda}} \frac{1}{\sqrt{kA}}$ I=1/R ٧ Current = electrical Polential $k = \frac{g \Delta x}{(T_1 - T_2) \Delta} = \frac{J/S \times m}{e C \times m^2}$ 88 electric. ٢ resistance ۲ = Top W/ Cm JS-j Watt $\frac{1}{R-\frac{4\pi}{kA}}g$ ۲ ĝ WOFM ٢ ۲ R = L/RA ۲ 1

Convection Newton - Rickman's law Heat fluer & Jemp: difference - Too K air $\frac{\delta}{A} \propto (T_{w} - T_{\infty}).$ Jet Tw > Two $\frac{\beta}{A} = h_{c} (T_{w} - T_{o})$ proportionality at the surface the velocity of air and as we more is slowly velocity increases-٢ 3 ٢ → film co-efficient **)** $g = h_c A (T_w - T_{oo})$ () | - E (*) Units = J/s m² ° C = w/m2°C = W/m²°F ٢ = W/m²K $Tw - T_{\infty}$ Q. (1/hA) - thermal resistance

Ti 8 8 18 To ho R &= hitrand ()) | hi <u> (</u> 1 8, = Ti-11 1/1.A $Q_i = hiA(T_i - T_i) \Rightarrow$ 8 $\theta_{2} = \frac{(T_{1} - T_{2})}{\ell \delta/k \Delta}$ $\frac{Q_1}{hiA} = T_1 - T_1$ 8 T1 = Ti - 91/4:A = $Q_2 = ho A (T_2 - T_0)$ $\frac{.83}{hoA} = T_2 - T_0$ È $Q = Q_1 \bar{\phi} Q_2 \bar{\phi} Q_2$ 33 $T_2 = T_0 + \frac{Q_3}{h_0 A}$ 1 Ì ٢ g = g1 = g2 = g3 Ś ٢ $= h_i A \left(T_i - \left(T_i - \frac{g_i}{h_i A} \right) \right) \stackrel{=}{\longrightarrow} \frac{T_i - \frac{g_i}{h_i A}}{h_i A} - \left(T_0 + \frac{g_3}{h_0 A} \right)$ ٢ ۲ L/kA all gave ۲ \overline{a} hoA $\left(.T_0 + \frac{a_3}{h_0A} - T_0 \right)$ Some 2001 Solve 2 get ٢ mywer $: hiA\left(\frac{g_{1}}{h_{1}A}\right) = Ti - \frac{g_{1}}{h_{1}A} - \left(T_{0} + \frac{g_{3}}{h_{2}A}\right)$ Q ۲ the A (103) ٢

 $g = \frac{T_i - T_1}{\frac{1}{1 - \frac{1}{1 - \frac{$ 1/ho A J: A 28 $T_{i} - T_{i} = \frac{Q}{h \cdot A}$ Fi-Fi BL K- To = 0 TC-TO = OF I + L + I hoA $g = \frac{t_i - t_o}{1 + \frac{L}{kA} + \frac{1'}{h_o A}} = \frac{\sqrt{zR_t}}{2}$ Overall heat transfer co-efficient (U)-> combining overall · co- efficient ef conduction + connection Q= UA(Ti-To) $Q = \underline{T_i - T_o}$ Unit = W/m² °C -0 1/UA compare 0.+ 2 1 - 1 + L + 1 VA hit kt hot $\frac{1}{k} + \frac{1}{k} + \frac{1}{k}$ U = X

412 Process Dynamics & Control \bigcirc Process Dynamis Stability Advanced Analysis Analysis control stability stealegies. Analysis Instrumentatio. It is the steedy of behaniour f of a perocers/ system when it is Process dynamics going from one steady state to another steady state 0 A process goes from one steady state to ramether steady ٢ state veben it is disturbed by some disturbance in a specific manner. It changed from one ۲ stendy state to another $q_{11}(\frac{m^{2}}{3}) = 10$ 20 in some time after q in is destuded. $\beta = 9 \text{ out }$ Girat stanlystule A How to understand the behaviour of places We will write the model of the perocers Model - It wears a set of mathematical equations which govern ethe phonen This mathematical equations are weitten by meeting the conservation daws for the process. Apart from the conservation equations the mathematical model also consists of constitutine delationships

constitutive relationships $e_n = PV = nRT$, $g = UA \Delta T_{invite}$, $(-v_A) = kG^m$ Mathematical model of a perocers is of two types:-i) Steady state model ii) Unsteady state model 27 we study the steady state behaviour than we will write steady state model and if we are interresed in S. unsteady state behaniour. than we will weite unsteady state model. Inpert / Disturbance / forcing functions . O Step input: when the change is redden and it continues, for infiniteitime $2(t) \int \chi_{SS} t < t_{\circ}$ $\chi_{SSTM} t \ge t_{\circ}$ rlt) Xss M 2.3 In Process dynamics, we are interested in the value of change and not the absolute values

to to - to = t* \bigcirc x(b) . [.M $\chi(t) - \chi_{ss} = \chi(t)$ jt*<0 X(t) { o M 't* ·≥ D In process dynamics we talk in terms of derivation variable & X(I) and not in terms of actual variables x(t). In process. 0 dynamics t=0 is that time at which the perocess is ۲ 1 disturbed. ۲ # 19(4) 755 Y(4) Q 9,2-10 O Im Im D ti 1.2m lm 0.2m. ٨ Tim t2 1.31 Im 0-3-۲ -> quent'= 10 y (t) -> actual output variable yos = 1 m ۹ $\chi_{ss} = 10$ $\chi(t) = \infty$ 0 y(t) = 1Yss ->. Steady state output X(J) = 20-10 Ú. = 10 Value Y(1) deviation output Noti * $Y(t)_{t=0} = 0$ variable Aquiation variable = actual variable - Initial Steady state • Step Eupet van be positive and negative in nature. · Step Ruput of magnitude M is written as X(it) = Mu(it) of the step input for negative step input of magnitude 4 B X(t) = - Mult)

Step Input if unit magnitude is called thit Step function (or) unit step input or its denoted by u(t). 2 Pulse / viectangular pulse. actual variable after shifting after shifting after 240 origin 255 X(t) IM D t=A $\begin{array}{ccc} x(t) & 0 & t < 0 \\ M & 0 \leq t \leq A \\ 0 & t' > A \end{array}$. The above function can be treated as a combination of two step functions. A pulse function is a combination of one positive step A negative step. (one tre step @ t=0 + one-re step X(t) = Mu(t) - Mu(t - A)t=A -ve sign . due to negative step If the alea of the is equal to 1 than it is called unit 4 pulse input.

3 3 Unit Empulse In a unit pulse function if A ->0 such that the area remains 1, the function becomes unit Impulse function. A unit Impulse fonction is denoted by X(t) = S(t)8 £ J-A Input (Ramp function (H) Kan Slope = M x(t) > (alt) X(t)= Mtu(t) I amp + siamp Step + step direction it **X**(±) 50 jo J clockins direction (t) 10 7 t

When a ramp input is given to another ramp input, than it leads to rotation. The direction of rotation depends upon the sign of the Input. of the anticlockunise -ve zs clockwise for 2nd case $Mt u(t) - M(t-t_i)v(t-t_i)$ X(#) = u(t-ti)qu(t)- gts just a representation (1) Sinvsoidal Input / function frequency & rad/s 3.
X (I) = A sin wit
Somplitude laplace Teansform laplace transform of a function f(1) is defined as 83 $L \left\{ f(t) \right\} = F(s) = \int e^{-st} f(t) dt$ ۲ $(D \ L \ \xi \ t^{u} \ \xi \ = \frac{n!}{S^{n+1}}; n \ should \ be + ve \ integer$

Methods to Calculate depriciation Book Value of the Equipment - defined as the value of equipment after the end of an year (financial year end or start of financial year) It is denoted by Va Service life - The time period up to rehich the use of the property is conomically feasible. eg- If the heat enclianger is putchagele for getting a find amount of head enchange, me j (equipment) veill not sieplace it with new one till the time it is Daning the noney, but the use is becoming costly than it gets supplaced with new one. That time of use is Jenouen as service life. If at is denoted by m Salvage value of Scrap Value -۲ The money abtainable by the selling of the equipment one 0 ۳ and above any charges is known as salwage value or Q2 Is uninstalling of trainsportation cost , reraf value. It is denoted by Vs If Vs is non - zero than this is known as Salwage Wake ۲ * I if <u>Vs</u> is zero that it is known as Sceap Value. *

Depriciation accounting has no relations with the real sale or purchase. This whole concept is made just for the sake of balance sheet. Straight line method assemptions. The defreciation amounts for the subsequent years are some Book Values $D_{l} = V_{0} - V_{1}$ June 51 7 NO D2 - V1-V2 YL = VI 03 = V3-V3 3.56 - 12 1.56 - 12 Dy 2 V3 - V4 IL 3 Vy Qu = Vn-1-Vs P $\frac{n}{2}$ \mathbf{D}_{i} \mathbf{V}_{0} $-\mathbf{V}_{2}$ 8 Salvage Service Sceap In the case of straight line, DI=D22 D3=. = Dm and that's why nD = Vo-Vs $D = \frac{V_0 - V_s}{m}$ 0 depricultio Ø Va = Vo - aD. Book Value

Straight line method is valid for both Salwage + Scrap. 3 8) Heat en is to be installed in a industry at an initial investiget of res. 10 Lake. The remain life of equipment was assumed to be 10 years + Salvage value approximite to be 2 lakh. If straight line depreciation method is used for deperied account the Cal the depreciation to boo during the second year of plant operation & book value after the \otimes end of third year operation \otimes . 8 Jakh. 63 80 thousand 5 $D = \frac{10 - 2}{2610}$ ٨ ۲ dependiation is ۲ always some for œ^ every year but $V_{a} = 10 - 3 \times \cdot 8$ 0 book value of is 7.6 Jakh not same for every year in steaight line method) Declining balance method assumption depreciation amounts. for the subsequent years are not some but 633 0 fined 1. factors are same deprisciation annous. ferrid V. forton Book Values 5 L Jime . $D_1 = 50 K$ ioy. J -1 107. I 02 2 45K 4.05L 2

4.5L =. 5L - 10% of 5L $V_1 = V_0 - f V_0$ $V_1 = V_0 (1-f)$ 4.05L = 4.5L - 10 Y. of 4.5L V2 2 V1 - + V1 = .V. (1-f) $V_2 = V_0 (1 - f)^2$ $V_{a=}$ $V_{o}(1-f)^{a}$ Da = Va-1 × f Vo, Vs + n these three data points is always aniable to us and it is supplied by the nonufactures. These data helps us in comparing different alternatives and also used in the dépriciation accounting purposes $V_{S} = V_{0} (1-f)^{n}$ $\frac{V_{S}}{V_{0}} = \left(1 - f\right)^{m}$ to (Ve) ra (Ve) n z (l-f) $f = 1 - \frac{v_s}{v_o} lln$

declining balance method is tialed only for Saluage @ and in the case of scrap the depricultion is 100 % in the first year which is not possible. Vo = 10 lakh Vs = 2 Jakh. 88 $f = 1 - \left(\frac{2}{10}\right).$ 8 8 f = 0.148 Va-1 = 10(1-0.148) Vg = 10 (1-0.148) 3 V2 = 7.259 Jakl. = 6.18 lateh. ۲ lg = 7.259. ×. 0:148 $V_{1} = 10(1-0.148)'$ 1.074 lakh. . = 8.52 D2 = 1.2 6 lable Double declining method 8 S assumption depriciation amounts for the subsequent years are not 82 Same but fined ' years are same 0 $V_a = V_0 \left(1 - f\right)^a$ & # Da = Va-1 × f/ f=2/m

f = 2/10 = 0.2 9) V32 V0(1-f)3 $V_{3} = 10(1-0.2)^{3}$ = 5.12 lakh. $V_1 = 10 (1 - 0.2)^{1}$ = 8 Jakh $V_2 = 10(1 - 0.2)^2$ - 6.4 labh. $D_2 = 8 \times D^2$ = 1.6 labh. Dz 2 6.4 x0.2 = 1.2.8 lakh. Surforme of the years digets method depreciation. amounts for the subsequent years are not same. Also, fined / plage factor is not same. $\theta_{a^{\pm}} = \frac{n - a + 1}{\Xi n} \left\{ V_0 - V_S \right\}$ $\left[\frac{\sum n = \frac{m(m+1)}{2}}{2}\right]$ Valid for Salvage as well as scrap. $\theta_{2} = \frac{10 - 2 + 1}{55 \sqrt{56}} \times (10 - 2)$ 13 6 stateto 1.3 labh.